

Schubklappe = pusher flap



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Coandajet with pusher -flap

Estimation and development status
July- 2018



Schubklappe = pusher flap



Who are we ?

News from South Nethersaxony

Bodo
Rengshausen Fischbach



Georg E.
Koppenwallner



Schubklappe = pusher flap



Who are we ?

News from the Solling



FFG Flugwissenschaftliche Fachgruppe Göttingen e.V.
Bunsenstr. 10 37085 Göttingen
<http://weperflieger-goettingen.de>

Weper Airfield near Moringen at the Solling

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Why new propulsion concepts for soaring?



- Take off – the daily problem

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Distributed Propulsion – Verteilte Antriebe

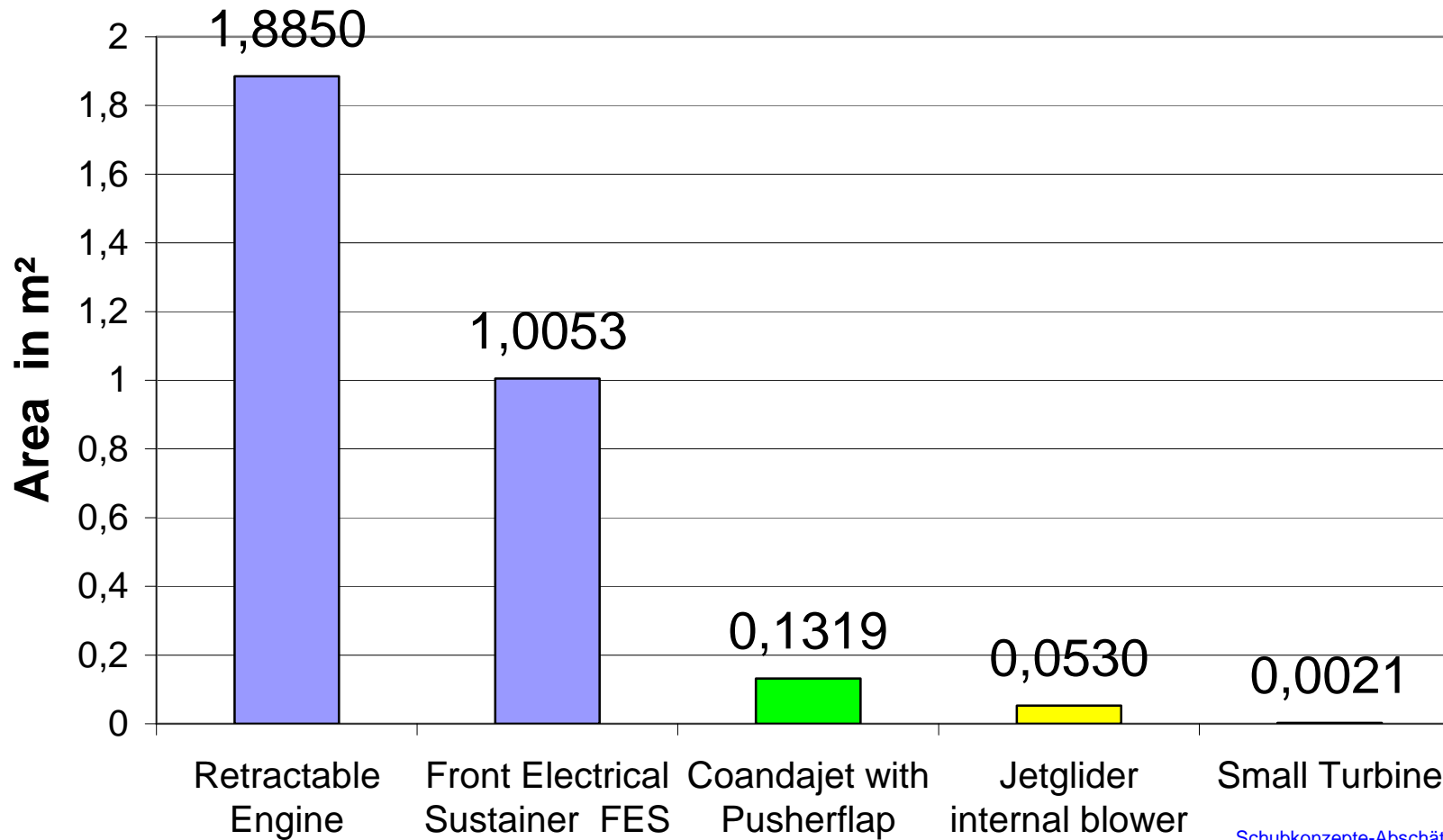


- Aerodynamic optimization by BLC = Boundary Layer Control

Schubklappe = pusher flap



Active surface of different propulsion concepts



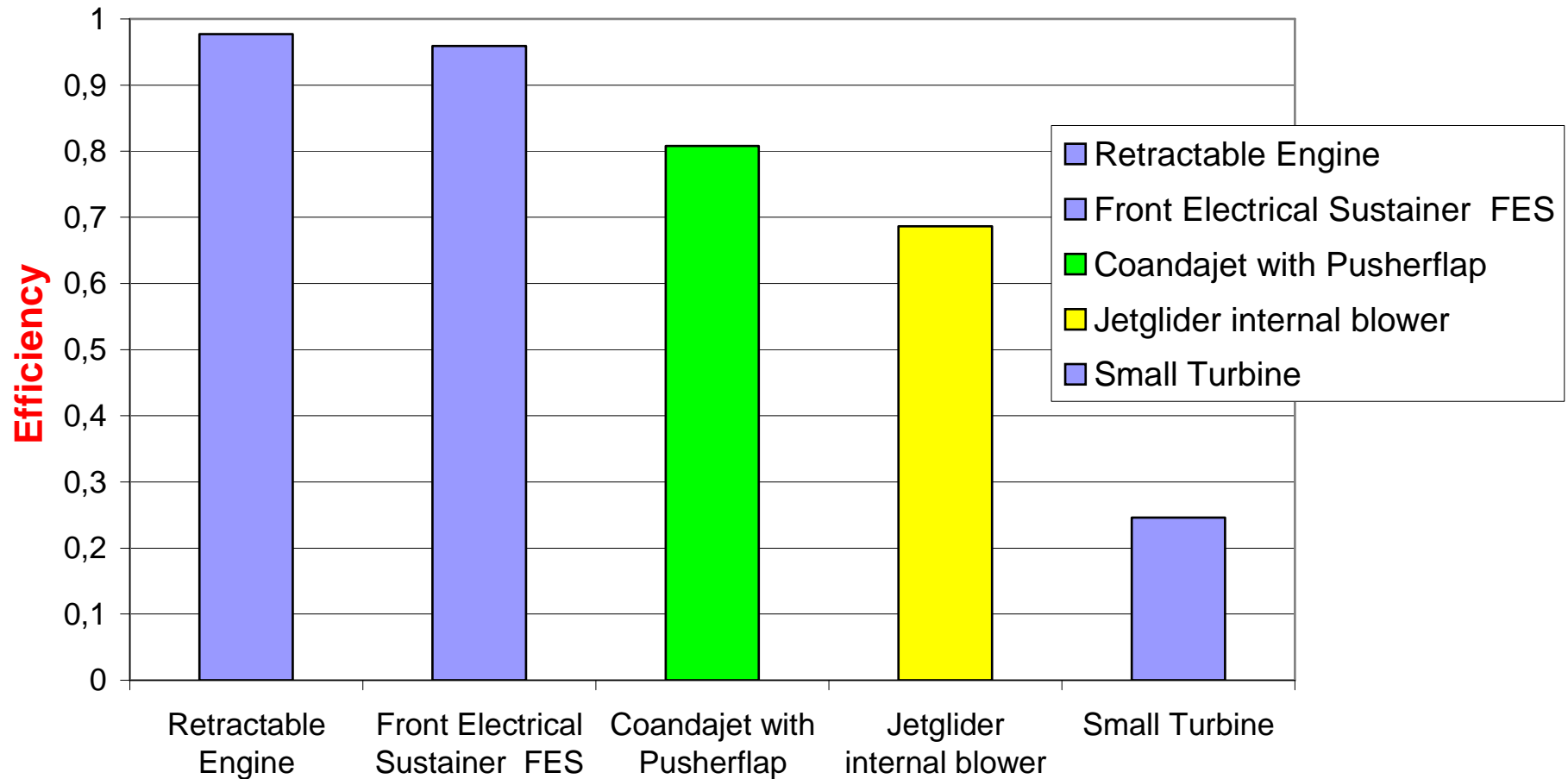
Schubkonzepte-Abschätzung-V15-SRU-OSTIV

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Propulsion efficiency of the propulsion efficiency

Propulsion Efficiencies to create a thrust of 100 N



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What is a pusher flap? / Was ist eine Schubklappe ?

Proposal:

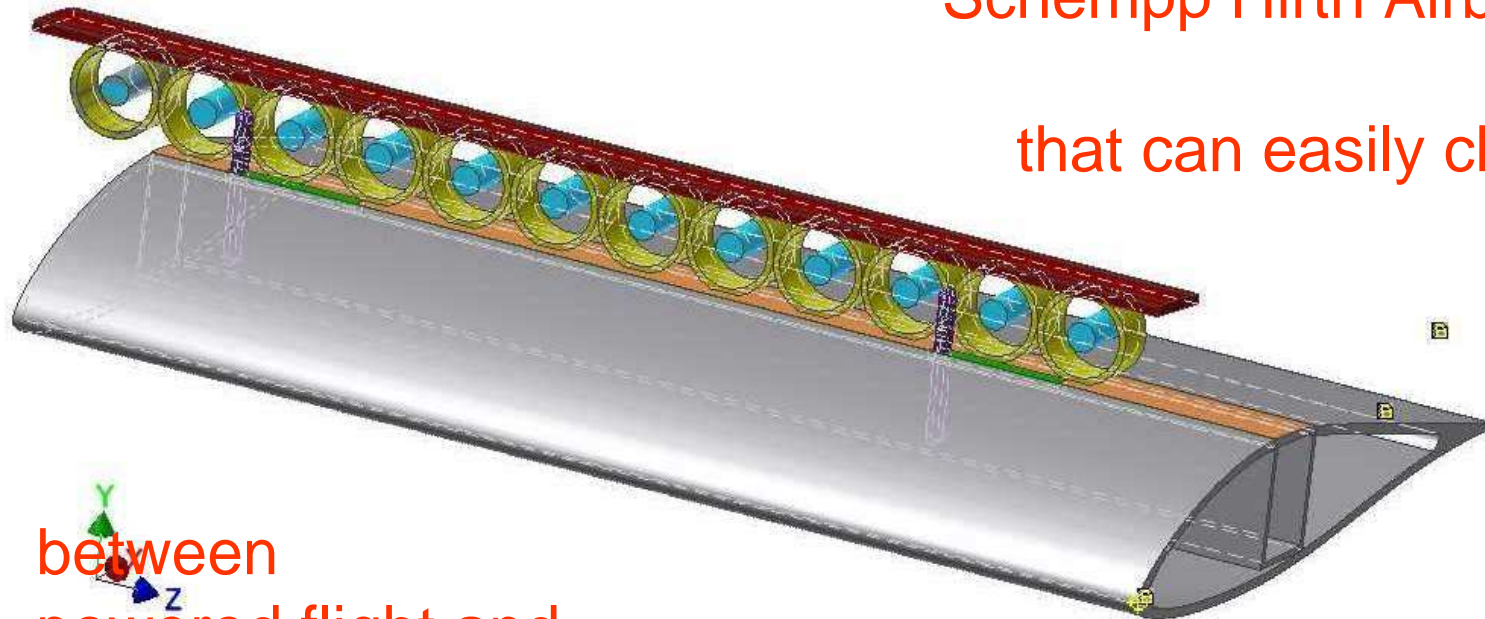
A device to extract a line of blowers out of the wings surface



What is a Coandajet ? / Was ist ein Coandajet ?

Proposal:

a glider with extractable
lines of thrust-blowers
instead of the
Schempp Hirth Airbrakes
that can easily change



between
powered flight and
soaring flight

Bild-Schubklappe-CAD-Segsym

What is the Coandaeffect?

jet-jet- and jet-wall interaction

N° 792.754

M. Coanda

Pl. unique

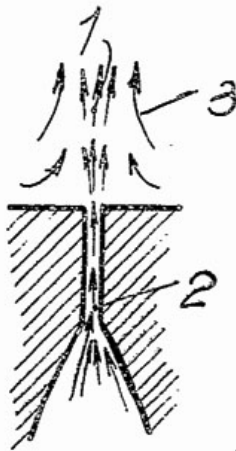


Fig. 1

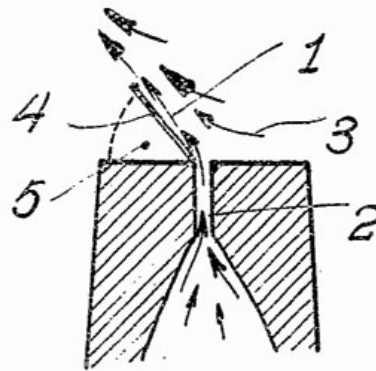
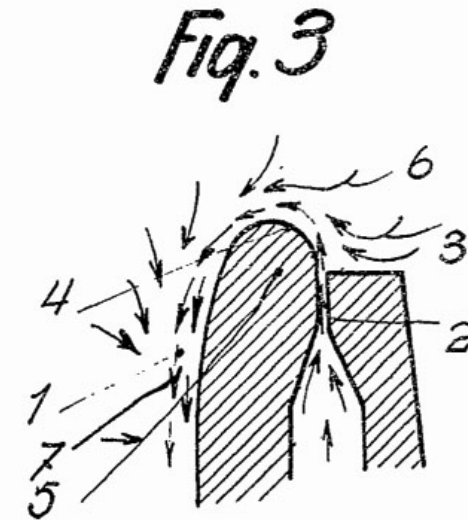


Fig. 2





Advantages of many impellers

- operation only with partial load during sustaining flight
- extrem high thrusts possible
- small impellers fit into the Schempp-Hirth flap
- no extra retraction elements for the blowers necessary
- redundancy
- lift increase in the region of the pusher flap

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Components off the shelf

- Impeller-provider – Hacker, Schübeler
- Batteries: Modellflug-Standard 8 kg/kWh Lipo
technical batteries LiFePo 12kg/kWh
- Controller : MGM Controller from Zlin near Brno or
YGE Controller from Wallenhorst
- Schempp-Hirth flap –a standard
- Soaring plane – in Europe available

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Components traditional - SH-flaps



Schempp-Hirth Klappen
St. Cirrus

Flugwissenschaftlichen
Fachgruppe Göttingen
DLR -Gelände

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Example: Schübeler Impeller DS 86-AXI HDS ,

m ~ 1,2 kg axial very short, < 110 mm with Intake

14 N/kW < thrust/power < 18 N/kW

$S_{\max} = 78\text{N}$



foto used with kindly permission of : Schübeler-Technologies

Link: <http://www.schuebeler-jets.de/de/>

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Beispiel: 16 Impeller für 700 N → 43,75 N pro Impeller

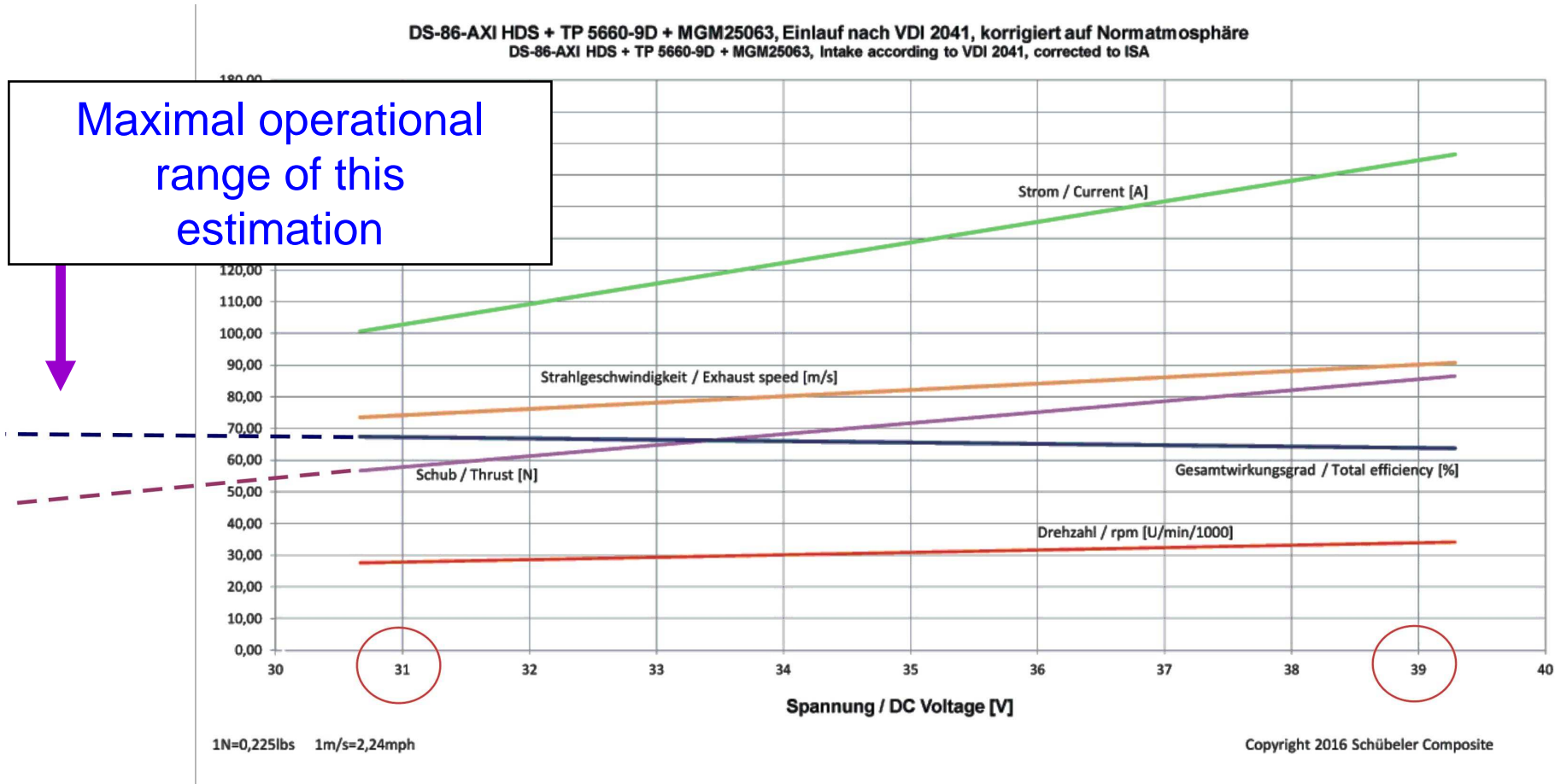


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Estimation after manufacturers specifications

Schubklappen-Schub-V33-17-SST Seglervariante-2018-07		Schübeler DS-86-AXI HDS + TP5660- 9D+MGM25083	
Voltage1	Spannung1	31	V
Thrust1	Schub1	58	N
Current1	Strom1	103	A
Power1	Leistung1	3193	W
Voltage2	Spannung2	39	V
Thrust2	Schub2	85	N
Current2	Strom2	155	A
Power2	Leistung2	6045	W
ThrustproPower1	SchubproLeistung1	18,16473536	N/kW
ThrustproPower2	SchubproLeistung2	14,06120761	N/kW
ThrustproPower-average	SchubproLeistunggemi	16,11297148	N/kW

Extrapolation

Estimation of thrust and power using reference values, Subskript 2, known values, Subskript X, extrapolated values

$$\frac{n_X^2}{n_2^2} = \frac{F_X}{F_2} \text{ und } \frac{n_X^3}{n_2^3} = \frac{L_X}{L_2} \Rightarrow F_X = \frac{L_X^{2/3}}{L_2^{2/3}} \cdot F_2$$

$$\Leftrightarrow F_X = L_X^{2/3} \cdot \frac{F_2}{L_2^{2/3}} \text{ mit : } \frac{F_2}{L_2^{2/3}} = T_{REF} \text{ : Referenzterm}$$

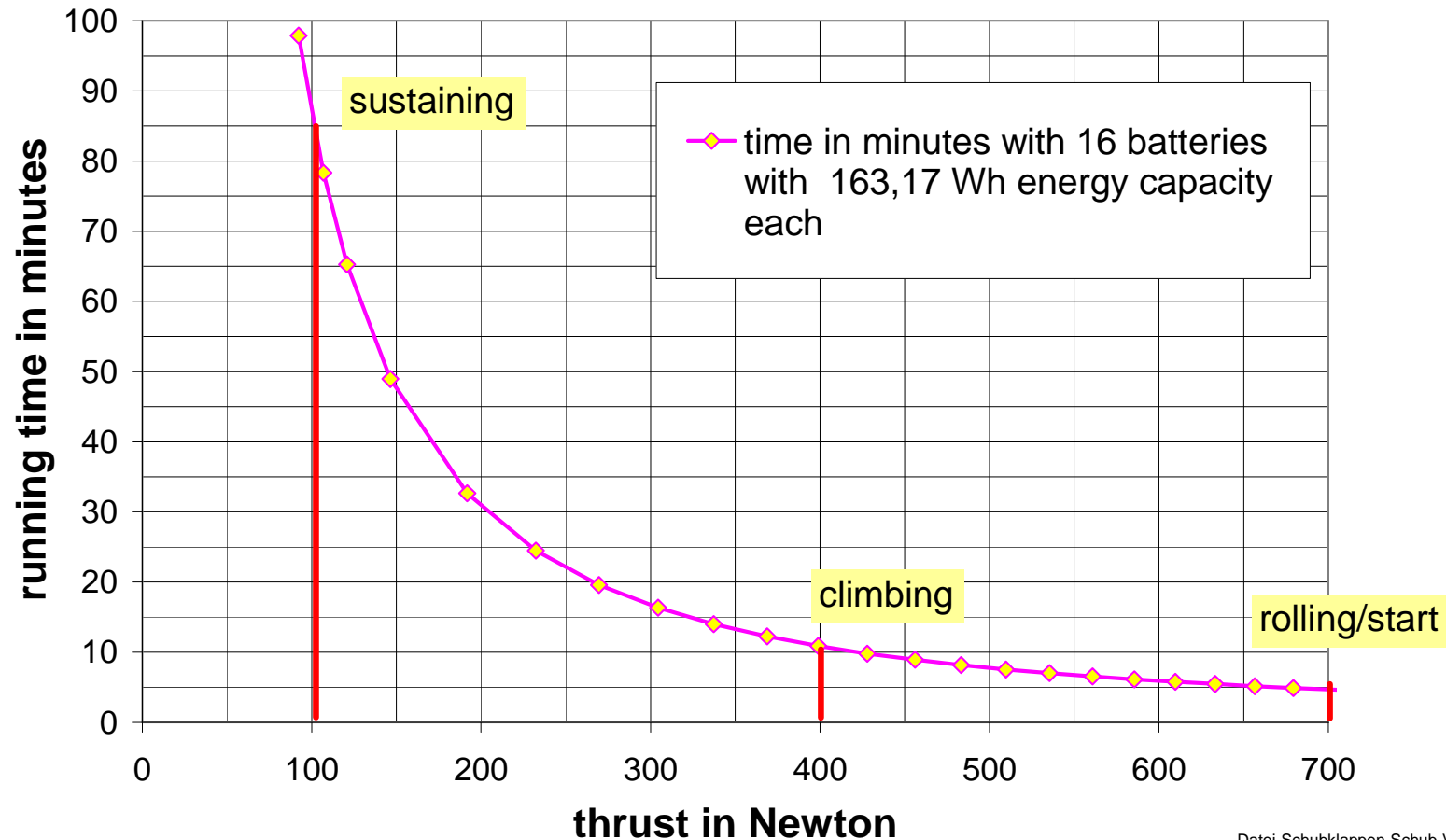
$$\Leftrightarrow F_X = L_X^{2/3} \cdot T_{REF}$$

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Estimation energy consumption $t = \text{Energy} / \text{Power}$

Batterie 7S, 6.3 Ah, $E(\text{Batt.}) = 7 * 3,7 * 6,3 \text{ VAh} = 163 \text{ Wh}$
6,4 kg/kWh - total 2,6 kWh



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Flight profil for a glider, rolling, climbing and sustaining flight

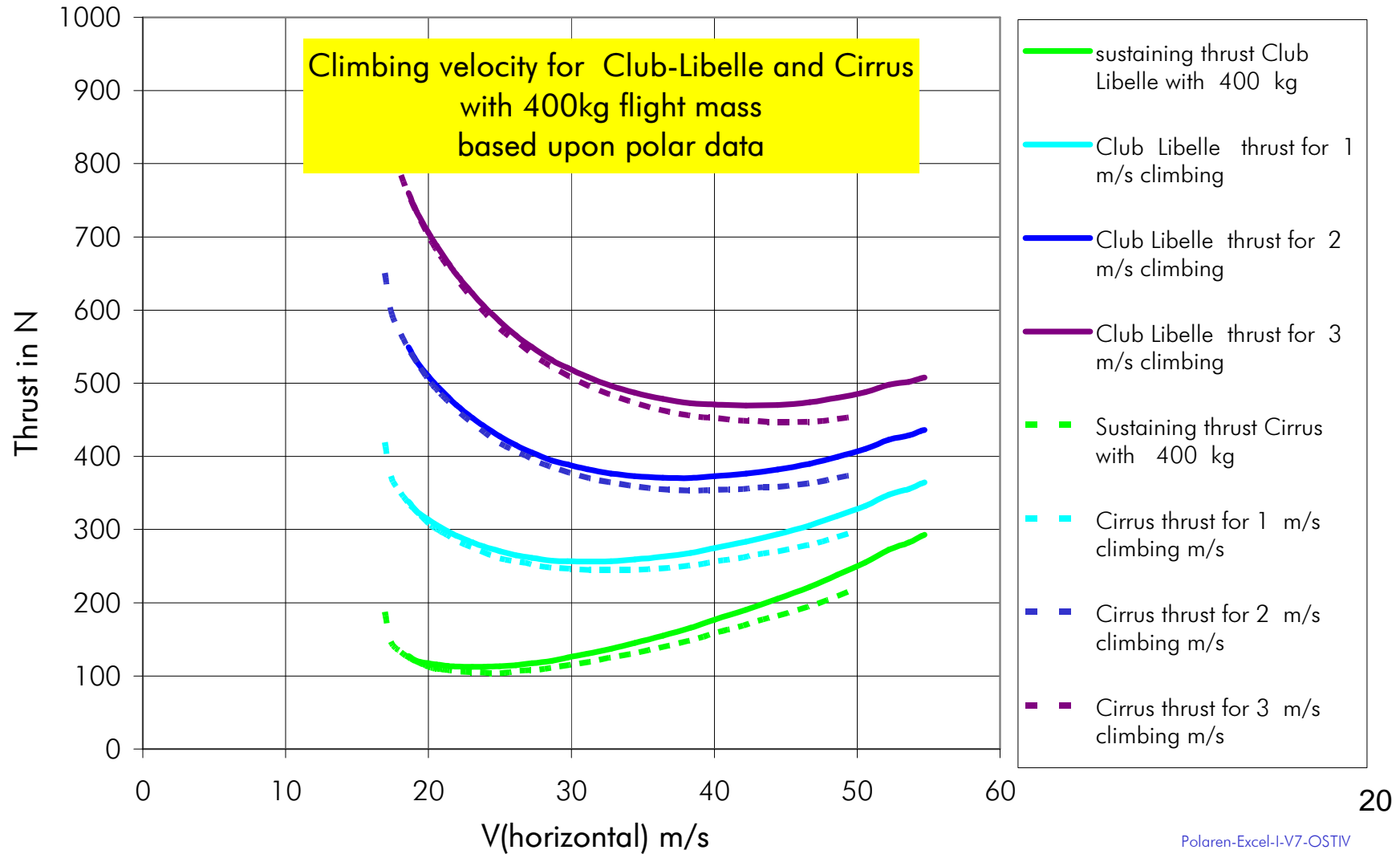
Weight impellers , 26 kg, $26\text{kg}/16 = 1,625 \text{ kg}$,
Weight batteries 17 kg Batteries, with 2.6 kWh
Flap mechanics 10 kg

flight profile	thrust in N	time in s	time in minutes	consumed energy kWh	consume d energy %	remaining energy %
rolling	700	60	1	0,59	22,47	77,53
climbing	400	300	5	1,33	51,07	26,46
sustaining flight	105	600	10	0,33	12,77	13,69

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Thrust and climbing, what is the benefit of 5 min 400 N ? 600 m altitude





Advantages and problems – Rethinking selflauch

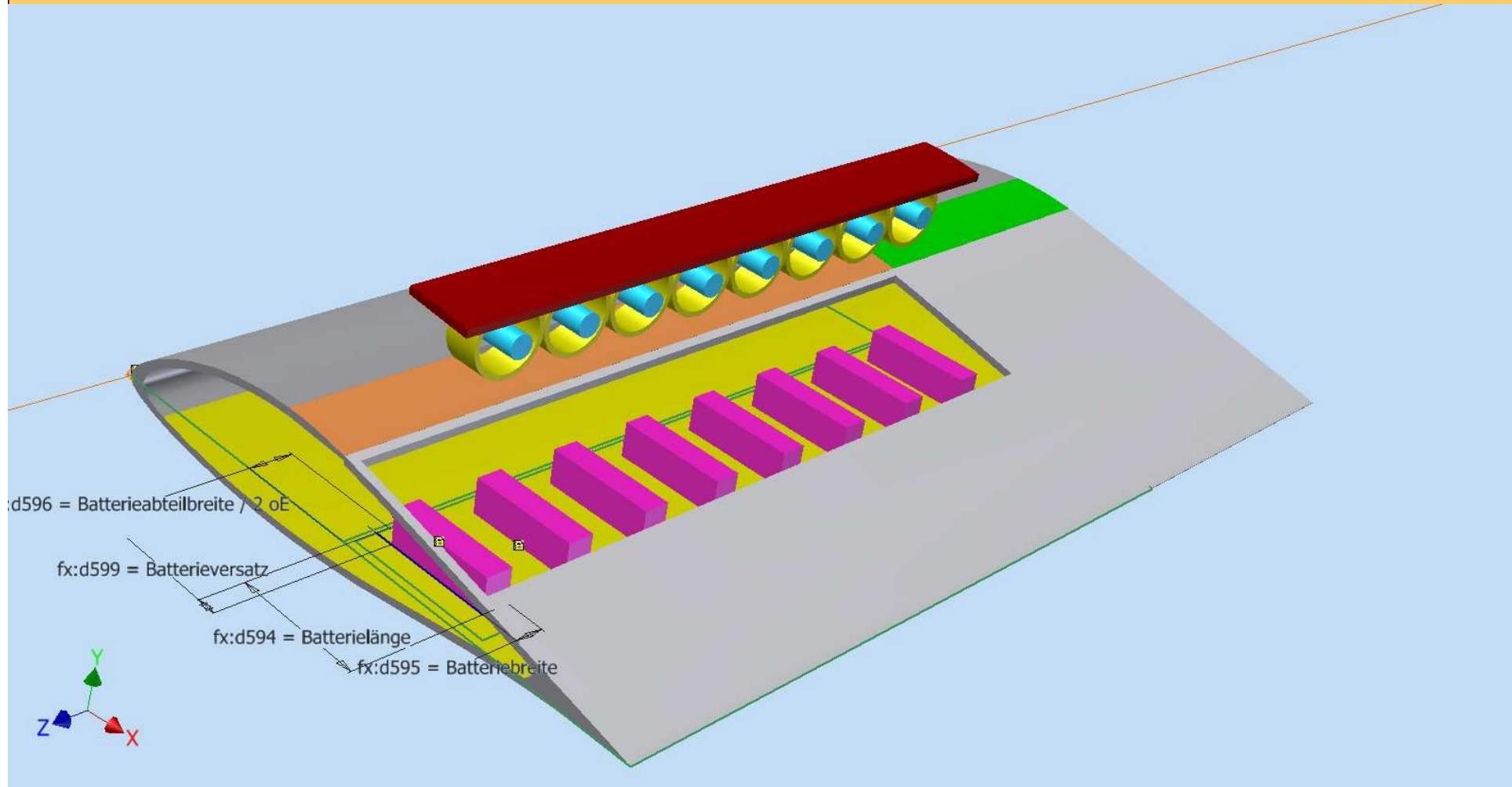
- Coandajet – increase of lift
- Power connection battery, controller and motor
- Extreme close distance between battery controller and motor!
- Accessibility of batteries, blower and controllers
- Fire protection many small batterie-units

- Profil thickness with carbon wings smaller compared to glasswings.
Increase of thickness along the flaps??
- Flap mechanics – with shear or linear ?
- Intake of blowers –how to optimize?
- Landing impact–sensitivity
- Use of pusherflaps, Schleppstrahl-landing?
- Charging device – central oder each wing separately?
- Controlling of the blowers – one signal for all controllers, one signal for each wing or a complete controllsystem comparable to multicopters or the volocopter

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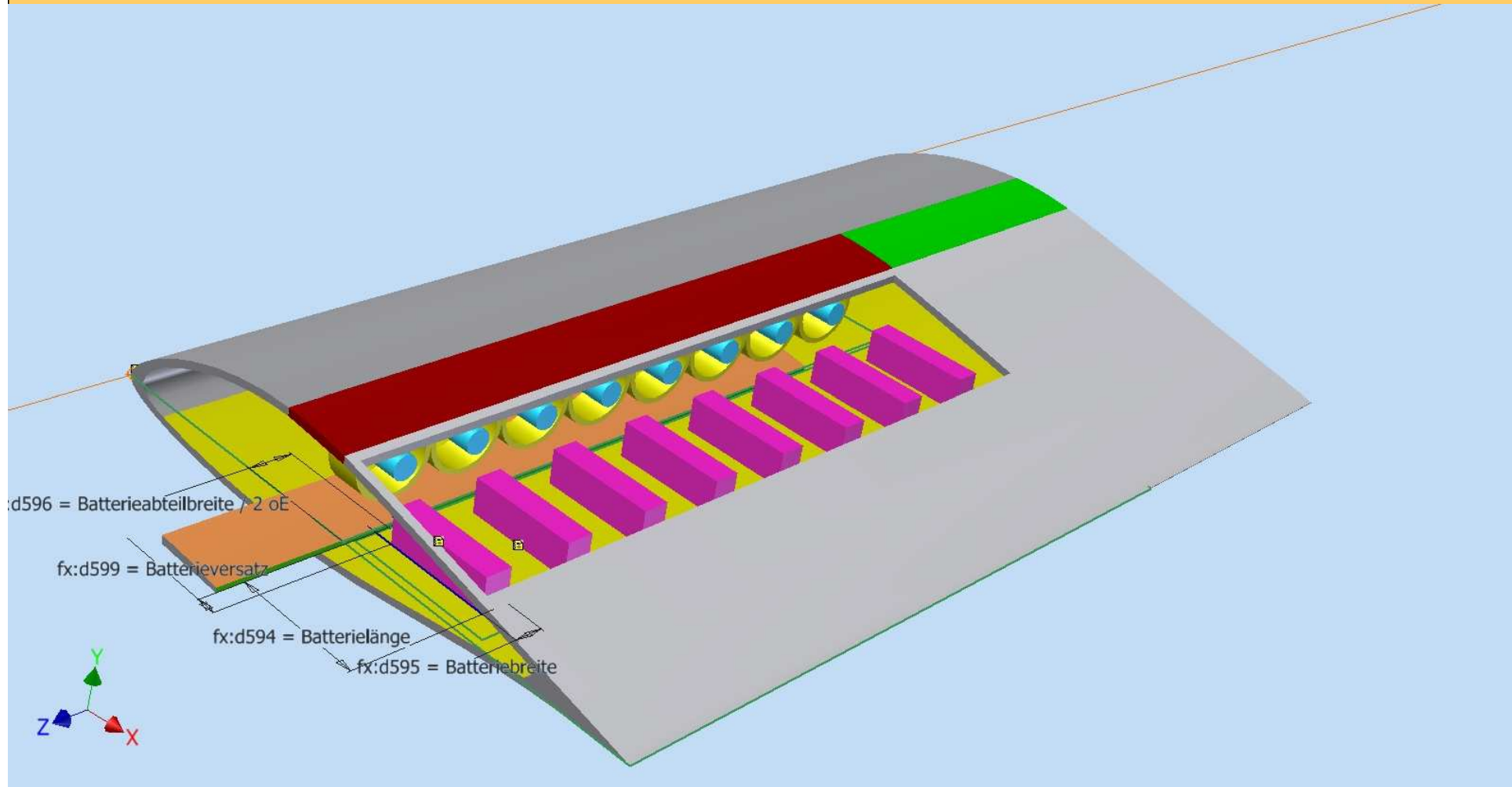
Battery assembly in the wing



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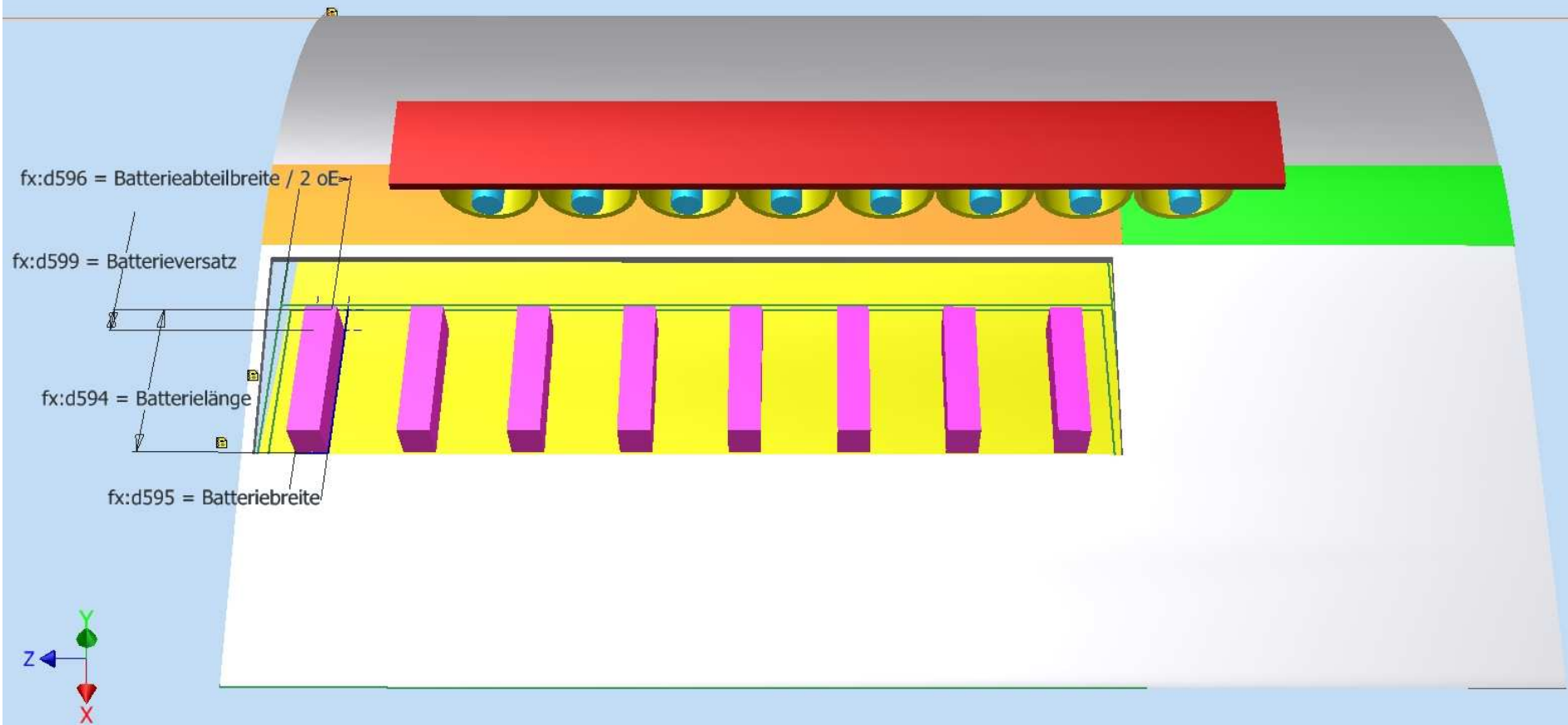
Battery assembly in the wing



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Battery assembly in the wing



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Summary

- Good ratio of risk and possible profit
- Need of a new unstressing glider-propulsion
- Self launch easy realizable,
- no modification of landing gear necessary
- Application also for other e-mobility concepts
- Limited modifications of existing types
- Need of a clima-neutral sport aircraft
- fits into the E-trend
- Forward looking but based on existing structures

Wasserkuppe

Joint development project:

- Research institutions
- Engineering offices and technology providers
- LTBs = Luftfahrttechnische Betriebe / Aeronautical companies
- Technology Management
- Akaflieds – academic flight clubs
- Glider manufacturer
- Battery manufacturer
- Private supporter

Brocken vom Acker her



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Other applications – Questions?



Coandajet.com – to go the green way

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Many thanks for your attention – Questions?



Coandajet.com – to go the green way

Lengderburg near Göttingen

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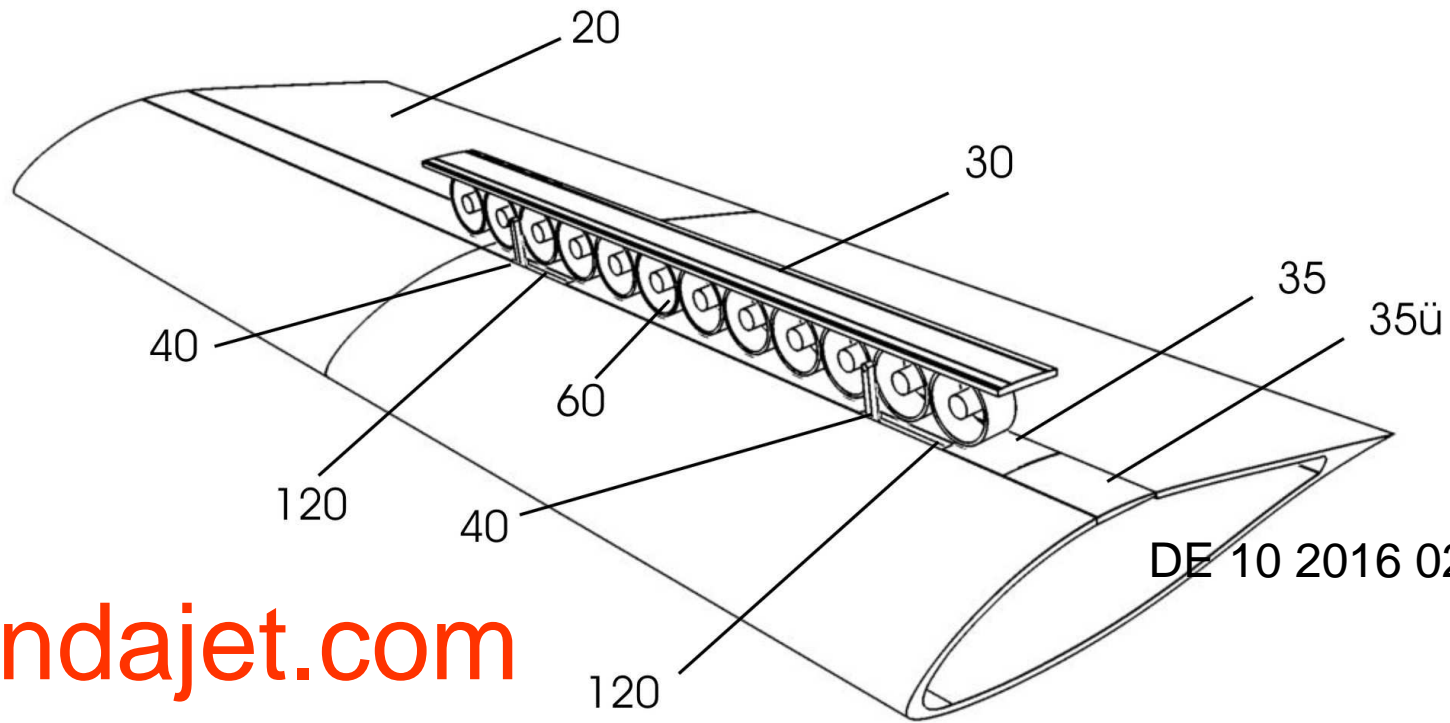


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Patent application DE 10 2016 020216.0

Fig. 11



Coandajet.com